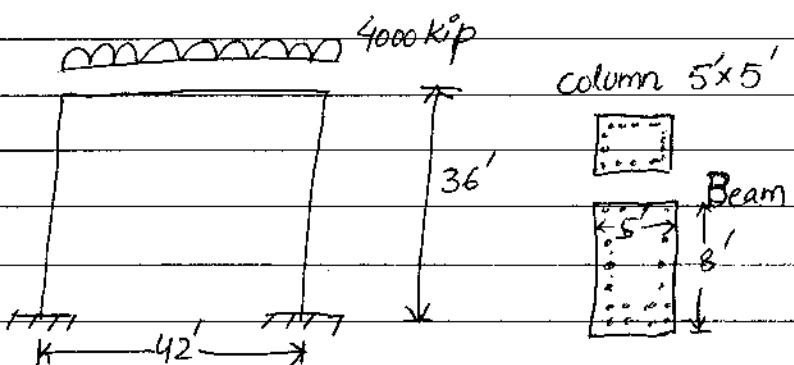


OpenSees : Open System for Earthquake Engineering Simulation.

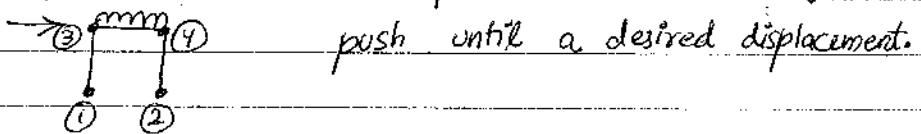
- The framework is a set of inter-related classes such as domains (data structures), models, elements, solution algorithms, integrators, equation solvers and databases.
- The modeling approach is very flexible.
- Wide range of solution procedures and algorithms.
- Fully programmable scripting language
- Three ways of execution of OpenSees/Tcl commands.
 - Input directly at prompt
 - Execute input file at OpenSees prompt (Source Inputfile.tcl)
 - " " " " MS-DOS prompt

Example :-



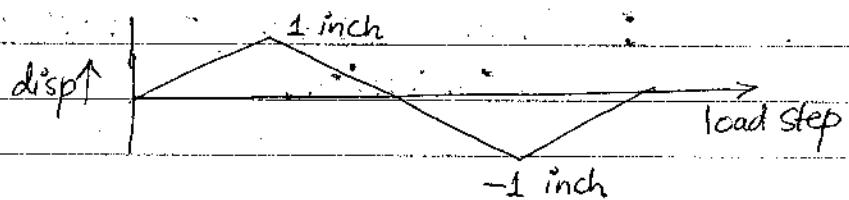
- Columns will be modeled as "elastic". (Can be replaced by more refined element models)
- Three different load cases.

1) Displacement controlled lateral pushover (static analysis)



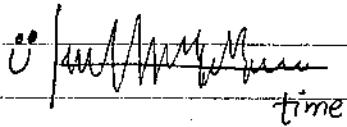
2) Displacement controlled reversed cyclic lateral loading (Static analysis).

A lateral load is applied at nodes 3 and 4 such that a predefined displacement history is achieved at node 3.



3) Dynamic ground motion input transient analysis.

A uniform $\ddot{\theta}$ history is imposed at all nodes constrained in a horizontal x -direction (1, 2)



Model Builder :-

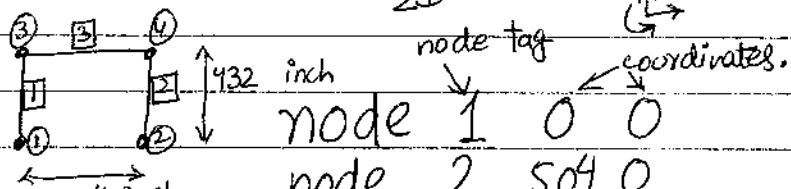
- defining model builder expands the Tcl command library to include OpenSees specific commands such as node and element definition.

ndm (no of dimensions)

ndf (no of dof per node)

model BasicBuilder -ndm 2 -ndf 3

Nodes :-



[for 3D]

node ---

node tag (x,y,z)

node 3 0 432

node 4 504 432

Boundary Conditions :-

0 = Unconstrained

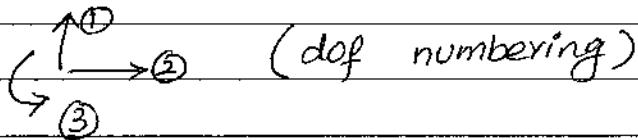
1 = Constrained.

In 3D,

eg fix 2 0 1 0 0 1 0

means

A node 2, axial elongation and torsion is restrained.



mass values corresponding to each nodal dof.

Mass :-

mass 3 5.18 0 0

can define

mass 4 5.18 0 0

or together also,

eg node 3 0 432 - mass 3 5.18 0 0

OR fix 3 0 0 0 - mass 3 5.18 0 0

example mass = 4000/2 kip

32 ft sec² × 12 in/ft

Elements :- we will use here "Elastic beam column Element"

for 2D,

element elasticBeamColumn-----

Tag Start End node A E I Identifier ←
for previously defined

Coordinate transformation
object.

But first,

no space

geom Transf Linear 1

used to construct a coordinate transformation

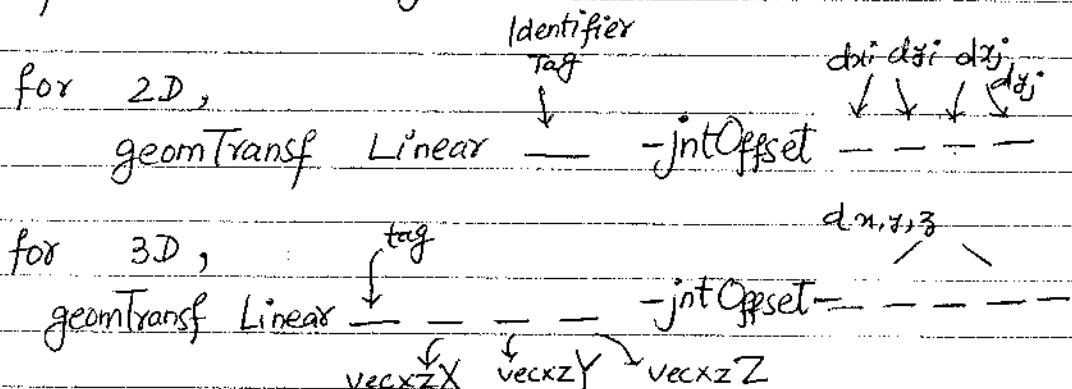
- It transforms beam element stiffness and object

resisting force from the basic system to the global coordinate system.

- Three types.

a) Linear Transformation :

Performs a linear geometric transformation of beam stiffness and resisting force from basic \rightarrow global



$d_{x_i}, d_{y_i}, d_{z_i}$ = Joint offset values

Absolute offsets specified w-r-t global for element-end node i .

Same for d_{x_j} etc \rightarrow node j .

$vecxzX, vecxzY, vecxzZ$ = X, Y, Z components of vector $vecxz$. \rightarrow The vector used to define the local

(opensees.berkeley.edu) $x-z$ plane of local.

opensees/manuals/usermanual/241.htm Local y axis = $(vecxz) \times (x\text{-axis})$
cross

These components are specified in the global coordinate system X, Y, Z and define a vector that is in a plane parallel to $x-z$ plane of a local coordinate system.

b) P-Delta Transformation

- Performs a linear geometric transformation of K and resisting force from basic \rightarrow global
Considering 2nd Order PΔ effects.

for 2D, geomTransf PDelta — \leftarrow Tag - intOffset ---
 for 3D, geomTransf PDelta ----- intOffset ---
 (same)

3) Corotational Transformation :-

- performs an exact geometric transformation

geomTransf Corotational - -intOffset ---
(Same as above 2)

For 3D element,

element elasticBeamColumn

For $2D \rightarrow TAG$ node node A E I_y Tag [Tag start end A E G J I_y I_z Tag]
 $i \quad j$ Transform for

element elasticBeamColumn 1 1 3 3600 4227 108000 Transform
1

element elasticbeamColumn 2 2 4 3600 4227 108000 1

element elasticBeamColumn 3 3 4 5760 4227 4423680 1

	Columns	Beam
Area	$(5 \times 12) (5 \times 12)$	$(5 \times 12) (8 \times 12)$
I_3	$\frac{(5 \times 12) (5 \times 12)^3}{12}$	$\frac{(5 \times 12) (8 \times 12)^3}{12}$

Recorders :-

used to define the analysis output

file name

Tags node numbers

start node end

a) recorder Node -file out -time -node ... -nodeRange -
-region - -node all -dof -
tag ↓ Tags dof numbers ↓ response type

b) recorder Element -file -out -time -ele --- -eleRange --
-region -- -ele all -- -----
eg force

In example,

```
Recorder Node -file Node3.out -time  
-dof 1 2 disp
```

Recorder Element - file Element1.out - time
- ele 1 force

Before "Recorders" → loads should be defined.
in global
↓ coordinates.

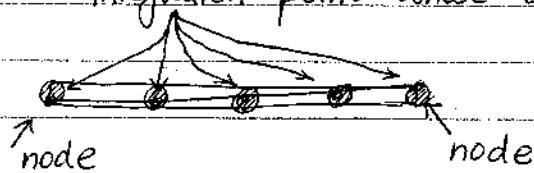
→ recorder Element -file ele1global.out -time -ele 1 globalForce

→ " " " ele1local.out " " " localForce

Sections:-

Section

integration point whose data is to be output.



Various responses eg,

- a) recorder Element -file mm.out -time-ele 1 Section 1 force
 b) " " " " " " " 1 Section 1 defor mation
 c) " " " " " " " " " " " " Stiffness
 " " " " " " " " " " " " stressstrain

Output format :- typically dependent on element type. ^{tag.}
general order,

Element	<i>globalForce</i>	2D, 3 dof	FX FY MZ
		3D, 6 dof	FX FY FZ MX MY MZ
	<i>LocalForce</i>	2D, 3 dof	$F_x F_y M_z$
		3D, 6 dof	$F_x F_y F_z M_x M_y M_z$

Section	force	F_x	M_x
deformation	axial strain	Curvature	
stressStrain	Stress	strain	

Beam-Column Elements

considers plasticity to be concentrated over specific hinge lengths.

Elastic Beam column
element

Nonlinear
beam column
element

displacement
with
hinges

- based on non-iterative or iterative force formulation
- Considers the spread of plasticity along the element.

Column element
element

distributed
plasticity

mass density
max no of iterations
- mass - iter -
Tolerance
Integration tag
Transformation tag
previously defined

element ^{non linear} dispBeamColumn

element tag start end node no of section
node integration points tag

Transformation tag

element dispBeamColumn ----- same as above ----- - mass -

element beamWithHinges

element tag inode hinge section
inode tag length tag hinge length
Section Length tag at node "i" "j" "j" length
previously defined
Corresponding to node "i"

E A I_z Transform density
tag tag tag

- mass - iter -

for 3D, add here

I_y G J

SECTIONS :-

for FD or $\sigma\epsilon$ relationships

at beam-column and plate sample points.

Section Elastic — E A I_z I_y G J

give any tag

F-D quantity to be modeled by this section object.

Section Uniaxial —

\downarrow tag \downarrow uniaxial material
 \downarrow Tag

P axial F-D
 M_z $M\phi$ about Z

V_y Shear F-D along Y

M_y $M\phi$ about Y

V_z Shear F-D $\rightarrow z$

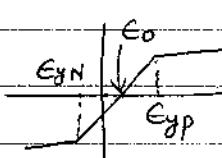
T Torsion F-D

uses a previously defined uniaxial material to represent a single section F-D response.

Uniaxial Materials :-

→ Uniaxial Material Elastic

→ " ElasticPP



? damping tangent
tag ↓ E ↓ optional, default
"0"

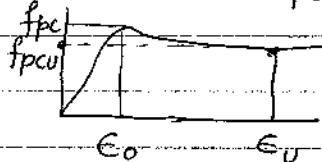
E_NP E_N E_0

→ " Plastic ENT

↓
elastic no tension.

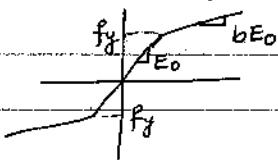
→ " Concrete01

↓ tag ↓ f_pc ↓ E_0 ↓ f_pcu ↓ E_u



→ " Steel01

↓ tag ↓ f_y ↓ E_0 ↓ b



previously
defined
material

Elements :-

→ element truss

or

" "

(tag ↓ i node ↓ j node ↓ A ↓ tag)

✓ previously
defined
section tag.

→ elasticBeamColumn

→ nonlinearBeamColumn

force based

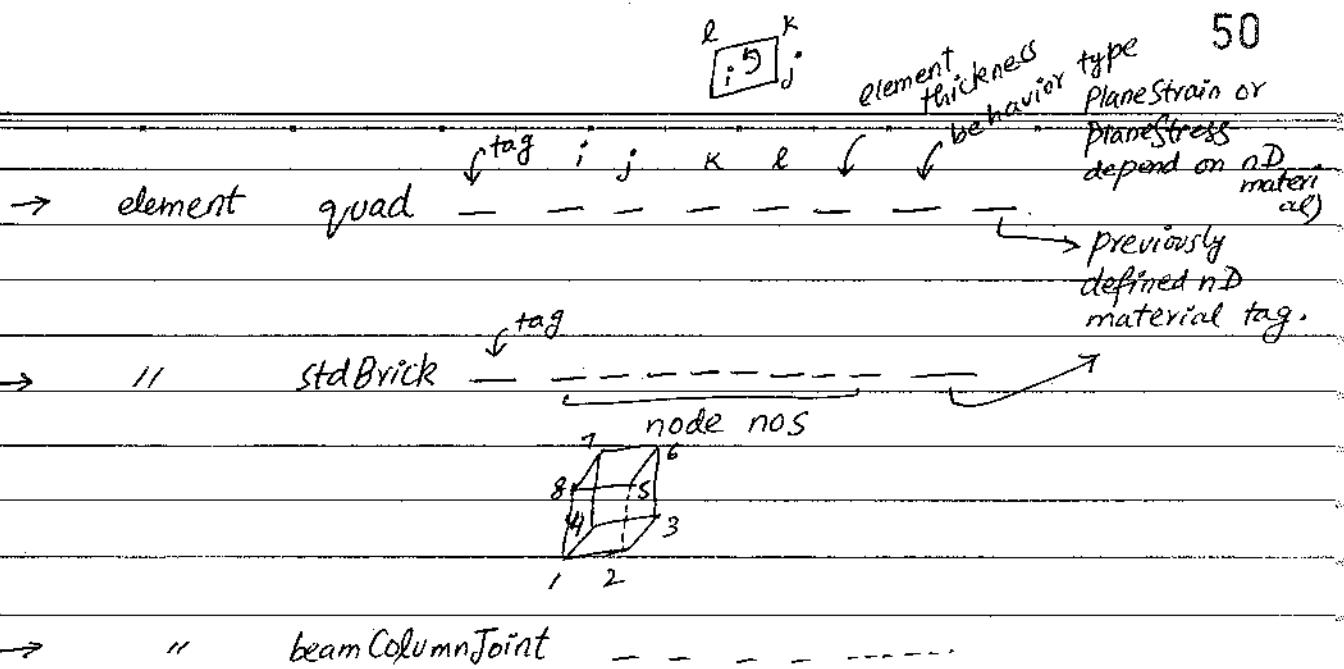
• distributed plasticity
(NonlinearBeamColumn)

• Concentrated plasticity with
elastic interior (beamWithHinges)

→ Displacement Based
(dispBeamColumn)

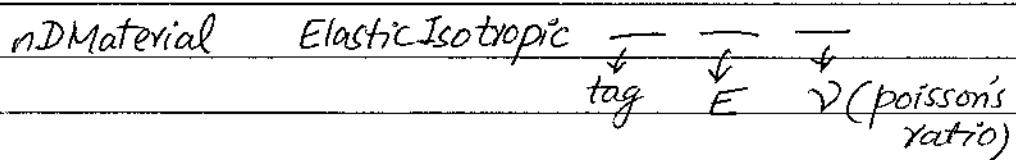
→ Flexure-Shear interaction displacement - based

beam - column element. (by Leo Massone) 2006.



nD Materials :-

Represents σ - ϵ relationships at the integration points of continuum and F-D elements.



Loads and Analysis :-

Applying loads is a 3-step process.

1) define load pattern

2) " Analysis

3) Apply loads when you execute analysis.

1) Load definition :-

a) Plain pattern, for

Load controlled nodal displacement

i) Nodal loads, gravity and lateral loads or

ii) Single point constraints (such as disp control at a node)
typically used for constant s
at a node

iii) Element loads, UDL

b) Uniform Excitation pattern U_g to all fix nodes

c) MultipleSupport pattern U_g to specified nodes

2) Analysis definition :-

for each analysis. 7 items required.

a) Constraints

- enforce a relationship between dofs enforced in analysis

b) Numbered

- determines mapping b/w equation #s and dof how dof's are numbered.

c) System

- Constructs Linear Solver to store and solve the system of equations in the analysis.

d) Test

- Certain solution algorithm require a convergence test at the end of iteration step.

e) Algorithm

- determines the sequence of steps taken to solve the non-linear equation.

f) Integrator:-

- determines the meaning of the terms in the system of equation object used for,

(i) determine the predictive step for time

(ii) Specify the tangent matrix and residual vector

(iii) determine the corrective step based on at any iteration
disp increment dU .

g) Analysis :

Three types available.

(i) Static Analysis \rightarrow Solves $KU = R$ without M, C matrices

(ii) Transient Analysis \rightarrow time dependent Analysis, time step = Constant

(iii) Variable Transient Analysis \rightarrow time step = Variable

3) Analysis execution :-

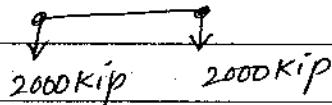
analyze —

↓
no of steps

GRAVITY LOADS :- → independent of the type of lateral loading and hence they are considered part of the structural model.

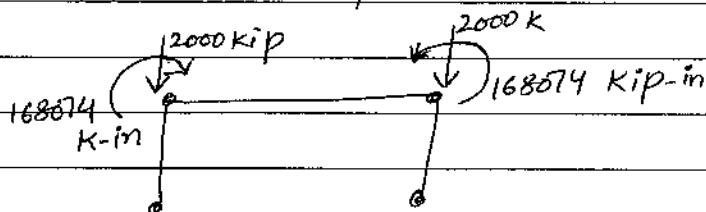
- nodal forces and moments.

$$\underline{F = 4000/2 = 2000 \text{ kip}}$$



$$M = \frac{wl^2}{12}, w = \frac{4000}{42 \times 12} = 7.94 \text{ kip/in}$$

$$M = 168674 \text{ kip in}$$



- Load pattern definition :- Plain pattern + Linear time Series

pattern plain ————— }
tag time series type

Load ————— }
node tag / / /

Load values corresponding to 3 dofs eg in 2D

In example,

pattern Plain 1 Linear }
load 3 0 -2000 -168674

load 4 0 -2000 +168674

}

CREATE ANALYSIS :-

constraints Transformation

number of RCM, Reverse Cuthill-McKee

System BandGeneral

algorithm

↳ banded system of equations

Test NormDispIncr 1.0×10^6

test NormDispIncr —

tolerance \rightarrow max number of

The test is applied to, iterations

$$K\Delta U = R$$

- Norm Unbalance checks $\sqrt{R^T R} < tol$
- Norm disp Increment checks $\| \Delta U \| < tol$
- Energy Increment checks $\frac{1}{2} (\Delta U)^T (R) K \Delta U < tol$

Algorithm Newton

→ The system of nonlinear equations in static analysis is of the form

$$R(U, I) = \lambda P^* - F_R(U)$$

Type of integrator depends on → static or dynamic
for static analysis,

$$\text{LoadControl} \quad \lambda_n = \lambda_{n-1} + d\lambda$$

$$\text{DisplacementControl} \quad U_j^n = U_{j,n-1} + dU_j$$

$$\text{MinUnbalDispNorm} \quad \frac{d}{d\lambda} (dU_n^T dU_n) = 0$$

$$\text{ArcLength} \quad dU_n^T dU_n + \alpha^2 d\lambda_n = ds^2$$

Integrator LoadControl 0.1

$$\overline{\Delta t} \rightarrow d\lambda_1$$

Analysis static

analyze 10

LoadConst - time 0.0

analyze $\overline{\Delta t}$ $\overline{\Delta t}$ $\overline{\Delta t}$ $\overline{\Delta t}$ $\overline{\Delta t}$

no of Δt Δt Δt Δt Δt

increments min max Jd

LATERAL LOADS - STATIC PUSHOVER :-

wipe

(clears any previous opensees objects)

pattern plain 2 Linear {

Load 3 100 0 0

Load 4 100 0 0

}

-Previously in gravity loads LoadControl integrator was used.
here DisplacementControl integrator will be used.

integrator DisplacementControl

↓ ↓ ↓
node dof du₁
tag tag

Load is applied to node 3

in the dir of dof 1

with a disp increment of 0.1, so

integrator Displacement Control 3 1 0.1

for 1inch Δ, 10 steps are needed

analyze 10

LATERAL LOADS - Cyclic Lateral Loads :-

pattern plain 3 Linear {

Load 3 100 0 0

Load 4 100 0 0

}

Load is applied to node 3 in the dir of dof 1
with $du_1 = 0.1$, for reversal $du_1 = -0.2 \rightarrow$ to amplitude
-1 so,

integrator DisplacementControl 3 1 0.1

analyze 10

integrator Displacement Control 3 1 -0.2

analyze 10

integrator Displacement Control 3 1 0.1

analyze 10

This can be put in to a foreach loop.

```
foreach Dincr {0.1 -0.2 0.13} {
    integrator DisplacementControl 3 1 $Dincr
    analyze 10
}
```

LATERAL LOADS - Dynamic ground motion :-

- from PEER strong-motion database,
download BM68elc.th (ground motion)
- A number of tcl scripts are available to the user at OpenSees website.

ReadSMDFile.tcl is a script procedure which parses a ground motion record from the PEER strong motion database by finding dt in the record header then echoing data values to the output file. The file should be saved in same directory as the OpenSees.exe.

- This file can be readin by the input file and used in analysis's procedure where an acceleration-time series is defined and used in a UniformExcitation load pattern.

Series -dt — — —
↓ ↓ ↓
\$dt for file — name of file — 1

pattern UniformExcitation — — —
↑ ↓ ↓
t/g direction time series
no type argument.

Create load pattern

source ReadSMDFile.tcl

ReadSMDFile BM68elc.th BM68elc.acc dt

set accelSeries "series -dt \$dt -filepath BM68elc.th
-factor 1;"

pattern UniformExcitation 2 1 -accel \$accelSeries

- Stiffness and mass-proportional damping factors

rayleigh — — — —

factors applied to

element/node mass element element element
matrix current initial committed
matrix stiffness matrix K K

Damping $\mathcal{D} = \alpha M + \beta K + \gamma K_{\text{current}} + \delta K_{\text{initial}} + \epsilon K_{\text{common}}$

set damping factors for math operations

rayleigh 0 0 0 [expr 2x0.2/pow(E
[eigen. 1],0.5)]

Various analysis components,

Create the analysis

wipe~~Analysis~~

Constraints plain

numberer RCM

System UmfPack

test NormDispIncr 1.0e-8 10

algorithm Newton

integrator Newmark 0.5 0.25

Analysis Transient

analyze [expr 10/0.02] 0.02

Time step does not have to be the same as
the input ground motion. no of time steps = $\frac{\text{Total duration}}{\text{time step}}$
 $= \frac{10}{0.02}$

PBD Seminar by Graham H. Powell

Session 1

Present Vs. Future

- a) Future methods are likely to be based on probability.
eg a limit state might be satisfied if there
only a 5% probability that its usage ratio > 1
- b) Future methods may also use performance measures
such as cost of repair ("dollars, downtime
and deaths") (different from deformations or hinge rotations).
- c) For the present, deterministic methods are enough
of a challenge.

Session 1 ends.

Tips and Tricks for Computer aided Structural Analysis

by
Siakat Basak

In earlier generation analysis programs, user had to supply
the programs the nodal coordinates, member incidences,
material and sectional properties and loads. + how the structure
is supported $\xrightarrow{\text{program calculates}}$ Member forces, nodal reactions,
joint displacements.

Now → draw your model on screen.

Linear Static Stress Analysis :-

Assumptions

- a) Loads applied slowly (no inertial or damping forces)
- b) Linearity assumptions $\xrightarrow{\text{if}}$
 - (i) All materials → hook's Law
 - (ii) Induced disp are small enough to ignore the change in K caused by loading.
 - (iii) BC do not vary during loading.