

**Assignment 1: Free Vibration Response of an SDF System**

**Question 1:**

Consider a simple structure which is idealized as a single-degree-of-freedom (SDF) system as shown in Figure 1. The mass  $m$  of the structure is  $4000 \text{ kg}$  which is supported by massless columns with combined stiffness  $k = 16 \times 10^5 \text{ N/m}$ .

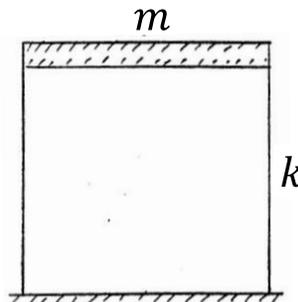


Figure 1: An undamped SDF system

Task 1: Determine the natural cyclic frequency " $f$ ", natural circular frequency " $\omega$ ", and natural time period " $T$ " of the system using analytical expressions derived in class.

Task 2: determine the change in the natural cyclic frequency " $f$ " of the system if

Case (a): the mass  $m$  is doubled

Case (b): the stiffness  $k$  is reduced to 50% of its original value

Task 3: Suppose that we want to reduce the natural cyclic frequency " $f$ " of this system to 85% of its actual value. For this purpose, suppose that only the mass  $m$  of this system is allowed to be changed. Determine the change in mass  $m$  which would result in the required reduction in natural cyclic frequency.

Task 4: Suppose that the real structure has some tendency to dissipate energy during the oscillations and we are interested in determining the damping coefficient  $c$  and critical damping ratio  $\xi$  of this system. For this purpose, a free vibration response test is conducted for this system and the resulting response is shown in Figure 2 below.

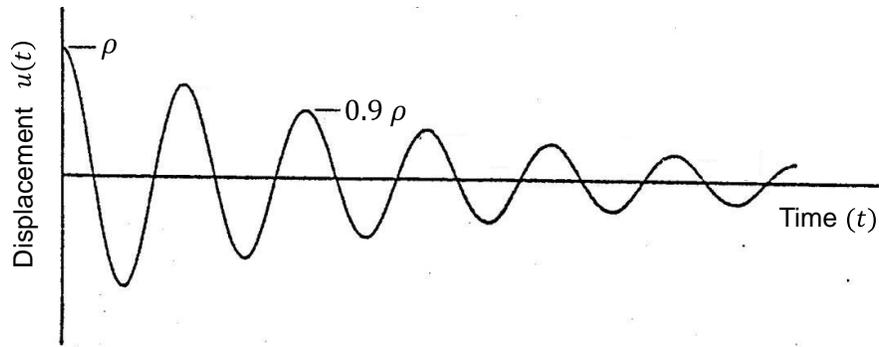


Figure 2: Free vibration response of the SDF system measured from experiment

Determine the damping coefficient  $c$  and critical damping ratio  $\xi$  of this system.

**Question 2:**

A simple structure is idealized as a single-degree-of-freedom (SDF) system and is shown in Figure 3 below. Its whole mass 12,000 Kg is assumed to be concentrated at top which is supported by two steel columns with hollow cross-sections. The columns are assumed massless, rigid in vertical direction, and firmly fixed to the rigid ground. Important structural dimensions and the column's cross-section are shown in Figure 3. The modulus of elasticity of steel is  $2 \times 10^{11} \text{ N/m}^2$ .

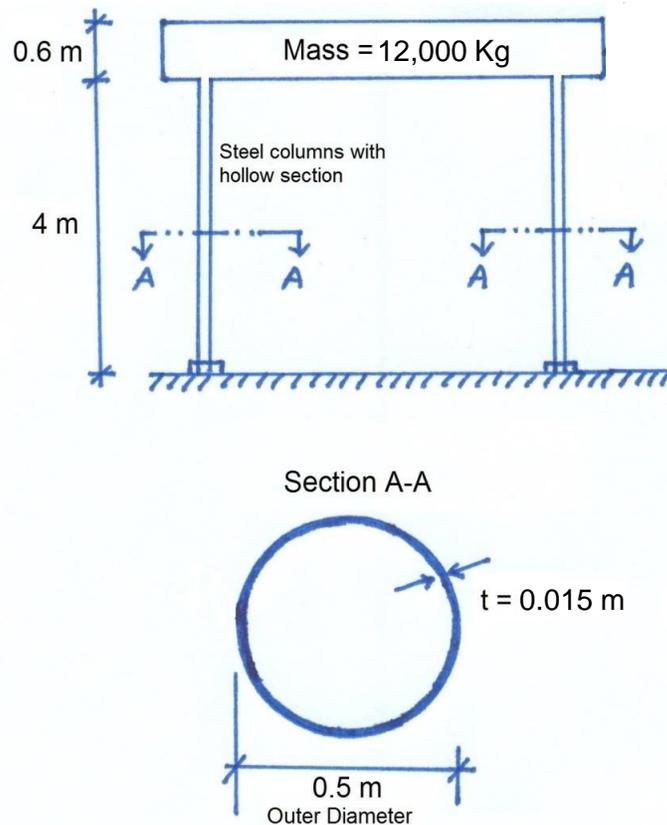


Figure 3: A simple structure supported by two columns and a lateral spring

Task 1: Construct a computer model of this simple structure using ETABS 2016 or SAP 2000. Determine the natural frequency ( $f$ ), natural circular frequency ( $\omega$ ) and the natural time period ( $T$ ) of this simple structure from computer software.

Task 2: Compare the values determined with computer program with analytical values determined using the expressions derived in class.

Task 3: Determine and plot the undamped free vibration response of the system for the following three initial conditions.

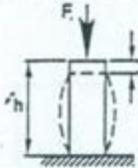
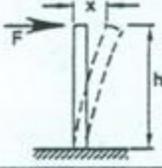
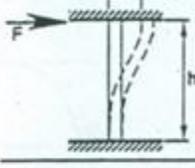
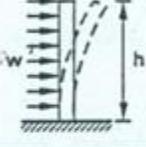
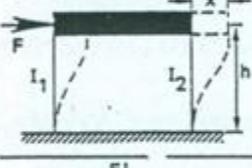
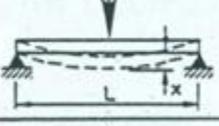
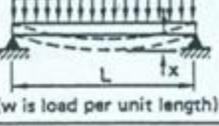
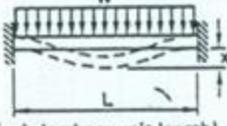
- a)  $u(0) = 0.03 \text{ m}, \dot{u}(0) = 0 \text{ m/s}$
- b)  $u(0) = 0.03 \text{ m}, \dot{u}(0) = 0.2 \text{ m/s}$
- c)  $u(0) = 0.03 \text{ m}, u(0.5 \text{ sec}) = -0.02 \text{ m/s}$

Overlay the plots of displacement responses for each case in the same figure.

Task 4: Consider the case when the simple structure shown in Figure 3 is modelled with an energy dissipation mechanism in the form of viscous damping. Suppose that the critical damping ratio ( $\xi$ ) of this structure is 0.05. Determine and plot the damped displacement response of structure for all three sets of initial conditions given in Task 3. Compare the displacement-vs-time plots for undamped and damped cases. Overlay the plots of displacement responses for both cases in the same figure.

## Elastic Stiffness

Deflection and stiffness for various systems (due to bending moment only)

System	Maximum Deflection ( $x$ )	Stiffness ( $k$ )
	$\frac{Fh}{AE}$	$\frac{AE}{h}$
	$\frac{Fh^3}{3EI}$	$\frac{3EI}{h^3}$
	$\frac{Fh^3}{12EI}$	$\frac{12EI}{h^3}$
	$\frac{wL^4}{8EI}$	$\frac{8EI}{L^3}$
	$\frac{Fh^3}{12E(I_1 + I_2)}$	$\frac{12E(I_1 + I_2)}{h^3}$
	$\frac{FL^3}{48EI}$	$\frac{48EI}{L^3}$
 (w is load per unit length)	$\frac{5wL^4}{384EI}$	$\frac{384EI}{5L^3}$
	$\frac{FL^3}{192EI}$	$\frac{192EI}{L^3}$
 (w is load per unit length)	$\frac{wL^4}{384EI}$	$\frac{384EI}{L^3}$